$$F = C_1 r v + C_2 r^2 v^2 \qquad F = \frac{mMG}{r^2} \qquad \mathbf{F} = \frac{d\mathbf{p}}{dt} = m\mathbf{a} \qquad dW = \mathbf{F} \cdot d\mathbf{r} \qquad a_{\text{cent}} = \frac{v^2}{r} = \omega^2 r$$

$$U = \frac{-mMG}{r} \qquad U = mgh \qquad U = \frac{1}{2}kx^2 \qquad K = \frac{1}{2}mv^2 \qquad K = \frac{1}{2}I\omega^2$$

$$E_{\text{tot}} = K + U = \frac{1}{2}mv^2 - \frac{mMG}{r} = \frac{-mMG}{2a} \qquad \mathbf{L} = \mathbf{r} \times \mathbf{p} \qquad I = \sum_{i} m_i r_i^2$$

$$m_1 r_1 = m_2 r_2 \qquad v = \omega r \qquad T^2 = \frac{4\pi^2 (r_1 + r_2)^3}{G(m_1 + m_2)} \qquad \omega = \frac{d\theta}{dt} \qquad \omega = \sqrt{k/m}$$

$$\mathbf{\tau} = \mathbf{r} \times \mathbf{F} = I\alpha = \frac{d\mathbf{L}}{dt} \qquad \alpha = \frac{d\omega}{dt} \qquad T = \frac{2\pi}{\omega} \qquad L = I\omega \qquad \mathbf{I} = \int_0^{\Delta t} \mathbf{F} \, dt = \mathbf{p_f} - \mathbf{p_i}$$

$$\omega = \sqrt{g/l} \qquad \omega_{\text{pr}} = \frac{\tau}{L_c} \qquad T^2 = \frac{4\pi^2 a^3}{GM}$$

Solid disk of mass M and radius R rotating about its cylindrical axis: $I = \frac{1}{2}MR^2$

Solid sphere of mass M and radius R rotating about an axis through its center: $I = \frac{2}{5}MR^2$

$$v_f - v_i = -u \ln\left(\frac{m_f}{m_i}\right) - gt \qquad I = I_{cm} + Md^2 \qquad I_z = I_x + I_y$$

$$f' = f\left(1 + \frac{v}{c}\cos\theta\right) \qquad \lambda' = \lambda\left(1 - \frac{v}{c}\cos\theta\right) \qquad \frac{dP}{dy} = -\rho g$$

$$PV = nRT \qquad R = 8.31\text{J/K} \qquad PV = NkT \qquad k = 1.38 \times 10^{-23}\text{J/K} \qquad N_A = 6.02 \times 10^{23}$$

$$\lambda = \frac{h}{p} \qquad E = hf = \frac{hc}{\lambda} \qquad \Delta L = \alpha L\Delta T \qquad \Delta V = \beta V\Delta T$$

$$h = 6.6 \times 10^{-34} \text{ J} \cdot \text{sec} \qquad \hbar = \frac{h}{2\pi} = 1.05 \times 10^{-34} \text{ J} \cdot \text{sec}$$

$$\frac{1}{2}\rho v_1^2 + \rho g y_1 + P_1 = \frac{1}{2}\rho v_2^2 + \rho g y_2 + P_2$$

Problem 7 (25 points)

Imagine a spherical, non-rotating planet of mass M and radius R. The planet has no atmosphere. A spacecraft of mass m ($m \ll M$), is launched from the surface of the planet with speed v_0 at an angle of 30° to the local vertical. The rocket burn is very short. Thus you may assume that when the spacecraft has a speed v_0 , it has not yet moved any appreciable distance.

a. (4) The speed v_{θ} is so high that the orbit is not bound. What is the minimum speed for which this is the case?

Now imagine that the orbit is bound and that in its subsequent orbit the spacecraft reaches a maximum distance of 15R from the center of the planet. At this distance the speed is V.

- b. (4) What is the ratio of v_0/V ?
- c. (4) What is the total energy of the spacecraft immediately after launch?
- d. (5) What is the total energy of the spacecraft when it is farthest away from the planet?
- e. (8) Write down one equation which would allow you to solve for v_0 in terms of M, G, and R (we are not asking you to solve this equation).

Problem 7 25 points

4 pts a)
$$\frac{1}{2}mv^2 = \frac{mMG}{R} \rightarrow v = \sqrt{\frac{2MG}{R}}$$

4 pts **b**)
$$mv_{_0}R\sin 30^{\circ} = mV(15R) \rightarrow \frac{v_{_0}}{V} = \frac{15}{\sin 30^{\circ}} = 30$$

4 pts c)
$$\frac{1}{2}mv_0^2 - \frac{mMG}{R}$$

5 pts d) same as in c. or
$$\frac{1}{2}mV^2 - \frac{mMG}{15R} = \frac{1}{2}m\frac{v_0^2}{900} - \frac{mMG}{15R}$$

8 pts e)
$$\frac{1}{2}mv_0^2 - \frac{mMG}{R} = \frac{1}{2}m\frac{v_0^2}{900} - \frac{mMG}{15R}$$

 $\frac{1}{2}v_0^2 - \frac{MG}{R} = \frac{1}{2}\frac{v_0^2}{900} - \frac{MG}{15R}$